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# Nonlinear stability and buckling analysis of composite functionally graded arches subjected to external pressure and temperature loading



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strain, stress, force and bending moment.

#### 1. Introduction

Arch elements are the basic units in civil infrastructures. The instability behavior of the arch element is common due to its high slenderness ratio, indicating the length is much higher than the dimensions of the cross-section. Therefore, it is significant to study the buckling responses of the arch structures. Existing studies [1–[7\]](#page-15-0) found that the buckling performance was induced by excessive hoop stress, which results from the mechanical loadings and/or thermal effects. When the arch buckles, the deformation may be symmetrical or asymmetrical as shown in [Fig. 1](#page-1-0) [8–[10\].](#page-15-1) In the present investigation, the symmetrical buckling mechanism is mainly concerned, and the asymmetrical buckling mechanism can also be evaluated with a similar derivation procedure. Following the simulation results [\[11,12\]](#page-16-0), the deformation is sensitive to the initial imperfection, indicating symmetrical initial imperfection results in symmetrical deformed shape and asymmetrical initial imperfection results in asymmetrical deformed shape, respectively.

Functionally graded materials (FGM) have been widely used in recent several decades. The concept of the FGM was proposed to dissipate the heat of the outer surface of the aircraft due to its high performance of thermal resistance [\[13\]](#page-16-1). And then, the FGM was extended in many other fields, such as nuclear power plants, thermal power plants and so on. In addition, the FGM was also used in the biological field for artificial bones or teeth, or in deep-sea apparatus for its high strength and corrosion resistance [\[14\]](#page-16-2). Many studies were concerned with the stability mechanism of the FGM arches. A typical FGM arch consists of two constituents with ceramic rich in outside surface and metal rich in inside surface as depicted in [Fig. 2](#page-1-1). Asgari et al. [\[15\]](#page-16-3) reported the buckling mechanism of the FGM arches under thermal effect. Based on the scheme of virtual displacement, the theoretical prediction of the critical temperature rise was obtained. It was concluded that material failure was prior to the thermal buckling. Bateni and Eslami [\[16\]](#page-16-4) studied the collapse mechanism of the shallow FGM arches under a radially-directed concentrated load. Both pinned-pinned and clampedclamped FGM arches were discussed and the buckling load was predicted by solving a set of nonlinear equilibrium equation. Swaminathan et al. [\[17\]](#page-16-5) presented the buckling mechanism of the FGM plates under different loadings types, boundaries, length-to-width ratios, and lengthto-thickness ratios. Simsek [\[18\]](#page-16-6) examined the two-dimensional Timoshenko FGM beams with the Ritz method. The critical buckling load is predicted by taking polynomial forms of the axial, transverse displacement and rotation of the beam cross-sections. Recently, Moita et al. [\[19\]](#page-16-7) reported the effect of material distribution on the buckling behavior of the FGM structures based on the theory of higher-order deformation. Five different types of FGM plates and panels were

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<span id="page-1-0"></span>

Fig. 1. A homogeneous arch deforms with (a) symmetrical, and (b) asymmetrical shapes.

<span id="page-1-1"></span>

Fig. 2. Material distributions of an FGM arch.

examined. Al-shujairi and Mollamahmutoğlu [\[20\]](#page-16-8) examined the buckling performance of FGM sandwich beams under the thermal variational field. The critical buckling load was obtained by using the generalized differential quadrature method.

All the above literature considered the arch was under a single load. In engineering practice, an arch may be subjected to two or more loading effects. Xi and Li [\[21\]](#page-16-9) investigated a semicircle FGM arch under uniform pressure in a thermal environment. The equilibrium paths were traced numerically. Pi and Bradford [\[22\]](#page-16-10) studied a homogeneous arch under uniform radial and thermal loading. The critical buckling pressure was obtained analytically and then verified by the numerical simulations. The pressure capacity sustained by the arch was sensitive to the temperature rise. Generally, the conventional approaches of developing the governing equilibrium equations were summarized as static eigenvalue method [\[23\]](#page-16-11) and energy method [\[24\]](#page-16-12). For the eigenvalue method, the geometric nonlinearities were neglected, resulting in an overestimation of the buckling load especially for the very shallow arches [\[25\].](#page-16-13) For the energy method, the principle of virtual work was commonly used as Refs. [\[15,16,22\].](#page-16-3) Nevertheless, it was difficult to obtain an explicit expression of the buckling load.

When an FGM arch is surrounded by a thermal rise field, a hoop compressive force develops significantly, and an outward displacement  $(w_T)$  generates as shown in [Fig. 3\(](#page-1-2)a). This thermal displacement  $(w_T)$ may result in different defomation and buckling load. Therefore, it is essential to explore the buckling, and the post-buckling behavior of the pressurized FGM arch under the thermal variational field.

The present work focuses on the nonlinear stability mechanism of the pressurized FGM arches in a temperature rise environment. Based on the deformed shape in [Fig. 3](#page-1-2)(b) [26–[31\],](#page-16-14) the analytical buckling pressure is derived for both clamped-clamped and pinned-pinned arches. Subsequently, the analytical solution is compared with numerical results for the FGM arch, as well as other closed-form predictions [\[22\]](#page-16-10) for the homogeneous arch, respectively. In the present numerical

<span id="page-1-2"></span>

Fig. 3. Deformed shapes of an FGM arch under (a) a temperature rise field, and (b) combined temperature rise field and pressure field.

model, the geometric nonlinearities are introduced and modified arclength algorithm is considered to track the pre- and post-buckling equilibrium paths. Finally, the effects of material properties and temperature rise on the internal forces and deformations are examined and discussed.

#### 2. Assumptions and formation of equilibrium equations

The geometric parameters are shown in [Fig. 4,](#page-2-0) with central angle *α* and its corresponding circumferential length *L*, where  $L = R\alpha$ , the cross-section  $b \times t$ , and the radius *R*, respectively. An orthogonal curvilinear coordinate system is defined, with  $x$ -,  $y$ - and  $z$ -axes in the tangential, axial, and radial directions, respectively. Two hypotheses are taken to facilitate the derivation process, yielding (1) the dimensions of the cross section  $b \times t$  are much smaller than the length (2*L*) and radius (*R*). Therefore, the cross-section remains in the plane during deformations; and (2) the material properties of the two constituents depend on temperature and only elastic properties are considered. The buckling is prior to the material failure in the present investigation.

As seen in [Fig. 4,](#page-2-0) the radial displacement is defined as  $w = w(z, \phi)$ and the tangential displacement is  $v = v(z, \phi)$ , respectively. Here  $s = R\phi$ , where  $\phi$  is the angle starting from the crown and works clockwise, and  $-1 \le \phi/\alpha \le 1$  for the right-half arch, which is employed for the analytical solution since loading and geometry are symmetrical to the vertical axis that passes through the mid-span as shown in [Fig. 4](#page-2-0). Following the thin-walled shell theory, the displacement components are expressed into

$$
v(z, \phi) = v_0(\phi) + \frac{z}{R} \frac{dw_0(\phi)}{d\phi}
$$
 (1)

$$
w(z, \phi) = w_0(\phi) \tag{2}
$$

where  $v_0(\phi)$  and  $w_0(\phi)$  are the tangential (circumferential) and radial displacements at the mid-surface of the arch, respectively. *z* is the distance from any point to the mid-surface in the radial direction. The

<span id="page-2-0"></span>

Fig. 4. Coordinate system and displacement definition of an FGM arch.

arch consists of two material mixtures as shown in [Fig. 2.](#page-1-1) The volume fraction of ceramic  $V_c$  and metal fraction  $V_m$  take the form of [\[15,16,21,32\]](#page-16-3)

$$
V_c = \left(\frac{1}{2} - \frac{z}{t}\right)^n
$$
 and  $V_m = 1 - V_c$  (3)

where  $n (n \ge 0)$  is volume fraction exponent. [Fig. 5](#page-2-1) shows the effect of volume fraction exponent  $(n)$  on the fraction of ceramic  $(V_c)$ . Following Voigt mixture theory, Young's modulus  $E(z)$  and coefficient of thermal expansion  $\lambda(z)$  vary in the thickness direction, yielding

$$
E(z) = E_m + (E_c - E_m) \left(\frac{1}{2} - \frac{z}{t}\right)^n
$$
\n(4a)

$$
\lambda(z) = \lambda_m + (\lambda_c - \lambda_m) \left(\frac{1}{2} - \frac{z}{t}\right)^n
$$
\n(4b)

where  $E_c$  ( $E_m$ ) is Young's modulus of the ceramic (metal), and  $\lambda_c$  ( $\lambda_m$ ) is the coefficient of thermal expansion of the ceramic (metal), respectively.

At any position of the arch wall, the total strain  $\varepsilon(\phi)$  consists of two parts: one is the strain at the mid-axis  $\varepsilon_0(\phi)$ , and the other is the strain induced by bending curvature, *zκ*, respectively. Therefore, we have

<span id="page-2-2"></span>
$$
\varepsilon(\phi) = \varepsilon_0(\phi) + z\kappa \tag{5a}
$$

$$
\varepsilon_0(\phi) = \frac{w_0(\phi)}{R} - \frac{1}{R} \frac{\partial v_0(\phi)}{\partial \phi} - \frac{1}{2R^2} \left[ \frac{\partial w_0(\phi)}{\partial \phi} \right]^2
$$
\n(5b)

The temperature rise  $(\Delta T)$  is introduced before the external pressure. The hoop strain induced by this thermal effect is

$$
\varepsilon_T = \lambda(z) \Delta T \tag{6}
$$

Similarly, the bending curvature is obtained from

<span id="page-2-1"></span>

Fig. 5. Effects of volume fraction exponent (*n*) on the ceramic fractions (*Vc* ).

$$
\kappa = \frac{w_0(\phi)}{R^2} + \frac{1}{R^2} \frac{\partial^2 w_0(\phi)}{\partial \phi^2}
$$
\n(7)

It is noted the underlined term in Eq. [\(5b\)](#page-2-2) is the increase of strain at the neutral axis due to bending curvature [\[23,31,33\].](#page-16-11) Therefore, this term is the geometric nonlinear term and significantly affects the buckling pressure [\[34\].](#page-16-15) Based on the one-dimensional thermoelastic scheme, the hoop stress yields

<span id="page-2-3"></span>
$$
\sigma(z, \phi) = E(z)[\varepsilon_0(\phi) + zx + \varepsilon_T]
$$
\n(8)

For the half arch, the total potential energy function is calculated by

$$
\Pi = \frac{1}{2} \iiint_0^V E(\mathbf{z})[\varepsilon_0(\phi) + \varepsilon_T + \mathbf{z} \mathbf{x}]^2 dV - W \tag{9}
$$

where *V* is the volume of half arch, and *W* is the work done by external pressure. Therefore

$$
W = Pb \int_0^L w_0(\phi) ds \tag{10}
$$

where  $s = R\phi$ , and *P* is the pressure loading, respectively.

### 3. Determination of critical buckling pressure

When an FGM arch is in a thermal rise environment, an upward displacement occurs. In such a case, external pressure on the arch will result in deformations as shown in [Fig. 6.](#page-3-0) Due to symmetrical deformations to line OC, only the right-half arch is examined. The radial displacement is assumed to be [\[22,26,29\]](#page-16-10)

$$
w_0(\phi) = \begin{cases} -(w_1 + w_2)\cos^2\frac{\pi\phi}{2\phi_0} + w_2; 0 \le \phi \le \phi_0 \\ w_2\cos^2\frac{\pi(\phi - \phi_0)}{2(\alpha - \phi_0)}; \phi_0 \le \phi \le \alpha \end{cases}
$$
clampe  
d – clamped ends (11a)

$$
w_0(\phi) = \begin{cases} -\left(w_1 + w_2\right)\cos^2\frac{\pi\phi}{2\phi_0} + w_2; 0 \le \phi \le \phi_0\\ w_2\cos\frac{\pi(\phi - \phi_0)}{2(\alpha - \phi_0)}; \phi_0 \le \phi \le \alpha \end{cases}
$$
 pinned – pinned ends

$$
(11b)
$$

The rotation and bending curvature are calculated by taking the first and second derivatives of  $w_0(\phi)$  to *s*, so we have

$$
\frac{1}{R} \frac{\partial w_0(\phi)}{\partial \phi} = \begin{cases} \frac{(w_1 + w_2)\pi}{2R\phi_0} \sin \frac{\pi \phi}{\phi_0}; 0 \le \phi \le \phi_0\\ \frac{-w_2\pi}{2R(\alpha - \phi_0)} \sin \frac{\pi(\phi - \phi_0)}{(\alpha - \phi_0)}; \phi_0 \le \phi \le \alpha \end{cases}
$$
clamped - clamped ends

$$
(11c)
$$

$$
\frac{1}{R}\frac{\partial w_0(\phi)}{\partial \phi} = \begin{cases} \frac{(w_1 + w_2)\pi}{2R\phi_0} \sin \frac{\pi \phi}{\phi_0}; 0 \le \phi \le \phi_0\\ \frac{-w_2\pi}{2R(\alpha - \phi_0)} \sin \frac{\pi(\phi - \phi_0)}{2(\alpha - \phi_0)}; \phi_0 \le \phi \le \alpha \end{cases}
$$
 pinned – pinned ends

(11d)

<span id="page-3-0"></span>

Fig. 6. Geometric parameters of half heated FGM arch under external pressure for (a) clamped-clamped, and (b) pinned-pinned arches.

$$
\frac{1}{R^2} \frac{\partial^2 w_0(\phi)}{\partial \phi^2} = \begin{cases} \frac{(w_1 + w_2)\pi^2}{2R^2 \phi_0^2} \cos \frac{\pi \phi}{\phi_0}; 0 \le \phi \le \phi_0\\ \frac{-w_2 \pi^2}{2R^2 (\alpha - \phi_0)^2} \cos \frac{\pi (\phi - \phi_0)}{(\alpha - \phi_0)}; \phi_0 \le \phi \le \alpha \end{cases}
$$
clampe  
d – clamped ends (11e)

<span id="page-3-1"></span>
$$
\frac{1}{R^2} \frac{\partial^2 w_0(\phi)}{\partial \phi^2} = \begin{cases} \frac{(w_1 + w_2)\pi^2}{2R^2 \phi_0^2} \cos \frac{\pi \phi}{\phi_0}; 0 \le \phi \le \phi_0\\ \frac{-w_2 \pi^2}{4R^2 (\alpha - \phi_0)^2} \cos \frac{\pi (\phi - \phi_0)}{2(\alpha - \phi_0)}; \phi_0 \le \phi \le \alpha \end{cases}
$$
\n
$$
\text{pinned} - \text{pinned ends} \tag{11f}
$$

<span id="page-3-5"></span>The curvature is continuous at  $\phi = \phi_0$ , so we get

$$
w_1 = \begin{cases} \frac{-\gamma^2 + 2\gamma}{(\gamma - 1)^2} w_2; & \text{clamped-clamped ends} \\ \frac{-2\gamma^2 + 4\gamma - 1}{2(\gamma - 1)^2} w_2; & \text{pinned-pinned ends} \end{cases}
$$
(12)

<span id="page-3-2"></span>where  $\gamma = \alpha/\phi_0$ . Here, the hoop strain is averaged through the arch span to obtain the equilibrium paths. Following the previous work [26–[31\]](#page-16-14), the averaged hoop strain *ε*¯ yields

$$
\bar{\varepsilon} \int_0^{R\alpha} ds = \int_0^{R\alpha} \left\{ \frac{w_0(\phi)}{R} - \frac{1}{R} \frac{\partial v_0(\phi)}{\partial \phi} - \frac{1}{2R^2} \left[ \frac{\partial w_0(\phi)}{\partial \phi} \right]^2 \right\} ds \tag{13}
$$

Noting that the following integration vanishes

$$
\int_0^{R\alpha} \left[ -\frac{1}{R} \frac{\partial v_0(\phi)}{\partial \phi} \right] ds = v_0(0) - v_0(\alpha) = 0
$$
\n(14)

Since  $v_0$  (0) = 0 at the crown and  $v_0(\alpha) = 0$  at the right ends for both clamped-clamped and pinned-pinned arches, respectively. With the introduction of Eqs.  $(11)$ ,  $(12)$  and  $(14)$  to Eq.  $(13)$ , we obtain

$$
\bar{\varepsilon} = \frac{w_2}{R} \tau_1 - \frac{w_2^2 \pi^2}{16R^2 \alpha^2} \tau_2
$$
\n(15a)

where

$$
\tau_1 = \begin{cases}\n\frac{\gamma^2 - \gamma - 1}{2(\gamma - 1)^2}; & \text{clamped-clamped ends} \\
\frac{1.273\gamma^3 - 1.819\gamma^2 - 0.181\gamma + 0.227}{2\gamma(\gamma - 1)^2}; & \text{pinned-pinned ends}\n\end{cases}
$$
\n(15b)

$$
\tau_2 = \begin{cases} \frac{\gamma^4 - 3\gamma^3 + 3\gamma^2}{(\gamma - 1)^4}; & \text{clamped-clamped ends} \\ \frac{4\gamma^4 - 12\gamma^3 + 12\gamma^2 - 3\gamma}{4(\gamma - 1)^4}; & \text{pinned ends} \end{cases}
$$
(15c)

<span id="page-3-3"></span>After substituting  $\varepsilon_0(\phi)$  by  $\bar{\varepsilon}$  in Eq. [\(9\),](#page-2-3) the energy function  $\Pi$  is expressed into

$$
\Pi = \begin{cases} \frac{R\alpha}{2E_1}\bar{N}^2 + \frac{\Gamma_3w_2^2\pi^4}{16R^3\alpha^3}\tau_3 - PRb\alpha w_2\tau_1; \quad \text{clamped-clamped ends} \\ \frac{R\alpha}{2E_1}\bar{N}^2 + \frac{\Gamma_3w_2^2\pi^4}{16R^3\alpha^3}\tau_3 + \frac{\pi(\Gamma_2\varepsilon + M_T)w_2\tau_4}{2R\alpha} - PRb\alpha w_2\tau_1; \\ \text{pinned-pinned ends} \end{cases}
$$
(16)

<span id="page-3-4"></span>where  $\bar{N}$  is the averaged hoop force, and written by

$$
\bar{N} = \Gamma_1 \bar{\varepsilon} + N_T \tag{17a}
$$

$$
\tau_3 = \begin{cases} \frac{\gamma^4}{(\gamma - 1)^4}; & \text{clamped-clamped ends} \\ \frac{\gamma^4}{4(\gamma - 1)^4}; & \text{pinned-pinned ends} \end{cases}
$$
(17b)

$$
\tau_4 = \frac{\gamma}{(\gamma - 1)}\tag{17c}
$$

Here,  $\Gamma_1$ ,  $\Gamma_2$ ,  $\Gamma_3$  are the stretching, stretching-bending and bending stiffness respectively, and  $N_T$ ,  $M_T$  are the thermal hoop force and bending moment respectively, and take the form of

$$
\Gamma_1 = b \int_{-t/2}^{t/2} E(z) dz = bt \left[ E_m + (E_c - E_m)/(n+1) \right]
$$
\n(18a)

$$
\Gamma_2 = b \int_{-t/2}^{t/2} E(z)z dz = \frac{nbt^2(E_c - E_m)}{2(n+1)(n+2)}
$$
(18b)

$$
\Gamma_3 = b \int_{-t/2}^{t/2} E(z) z^2 dz = \frac{bt^3}{12} \left[ E_m + \frac{(3n^2 + 3n + 6)}{(n+1)(n+2)(n+3)} (E_c - E_m) \right]
$$
(18c)

$$
N_T = b \int_{-t/2}^{t/2} E(z) \lambda(z) \Delta T dz
$$
  
=  $bt \Delta T$   

$$
\left[ E_m \lambda_m + \frac{E_m(\lambda_c - \lambda_m) + (E_c - E_m)\lambda_m}{(n+1)} + \frac{(E_c - E_m)(\lambda_c - \lambda_m)}{(2n+1)} \right]
$$
(18d)

<span id="page-3-6"></span>
$$
M_T = b \int_{-t/2}^{t/2} E(z) \lambda(z) \Delta T z dz
$$
  
=  $nbt^2 \Delta T \left[ \frac{E_m(\lambda_c - \lambda_m) + (E_c - E_m)\lambda_m}{2(n+1)(n+2)} + \frac{(E_c - E_m)(\lambda_c - \lambda_m)}{(2n+1)(2n+2)} \right]$  (18e)

#### 3.1. Clamped-clamped ends arch

Two unknowns,  $w_2$  and  $\gamma$ , exist in Eq. [\(16\).](#page-3-3) For the clampedclamped arch, the principle of minimum potential energy requires the first derivative of the total potential energy function  $\delta \Pi = 0$ , and the

<span id="page-4-0"></span>second derivative  $\delta^2 \Pi < 0$ , yielding

$$
\delta \Pi = \frac{\partial \Pi}{\partial w_2} \delta w_2 + \frac{\partial \Pi}{\partial \gamma} \delta \gamma = 0 \rightarrow \frac{\partial \Pi}{\partial w_2} = 0 \text{ and } \frac{\partial \Pi}{\partial \gamma} = 0
$$
(19a)

<span id="page-4-5"></span>
$$
\delta^2 \Pi < 0 \tag{19b}
$$

The first derivative of the averaged hoop force yields

$$
\frac{\partial \bar{N}}{\partial w_2} = \Gamma_1 \left( \frac{1}{R} \tau_1 - \frac{2w_2 \pi^2}{16R^2 \alpha^2} \tau_2 \right)
$$
(20a)

$$
\frac{\partial \bar{N}}{\partial \gamma} = \Gamma_1 \left( \frac{w_2}{R} \frac{d\tau_1}{d\gamma} - \frac{w_2^2 \pi^2}{16R^2 \alpha^2} \frac{d\tau_2}{d\gamma} \right)
$$
(20b)

<span id="page-4-1"></span>Therefore, two nonlinear equilibrium equations are obtained from Eq. [\(19a\),](#page-4-0) yielding

$$
R\bar{N}\left(\frac{1}{R} - \frac{w_2\pi^2}{16R^2\alpha^2} \frac{2\tau_2}{\tau_1}\right) + \frac{\Gamma_3 w_2\pi^4}{16R^3\alpha^4} \frac{2\tau_3}{\tau_1} = PRb
$$
\n(21a)

<span id="page-4-2"></span>
$$
R\bar{N}\left(\frac{1}{R} - \frac{w_2\pi^2}{16R^2\alpha^2}\frac{d\tau_2}{d\gamma}/\frac{d\tau_1}{d\gamma}\right) + \frac{\Gamma_3w_2\pi^4}{16R^3\alpha^4}\frac{d\tau_3}{d\gamma}/\frac{d\tau_1}{d\gamma} = PRb
$$
\n(21b)

<span id="page-4-3"></span>Substituting Eqs. [\(21a\)](#page-4-1) into [\(21b\)](#page-4-2) to eliminate the term *PRb*, the averaged hoop force takes an alternative form of

$$
\bar{N} = \frac{\Gamma_3 \pi^2}{R^2 \alpha^2} \Phi_1 \tag{22a}
$$

where

$$
\Phi_1 = \left(\frac{2\tau_3}{\tau_1} - \frac{d\tau_3}{d\gamma}\Big)\frac{d\tau_1}{d\gamma}\right) / \left(\frac{2\tau_2}{\tau_1} - \frac{d\tau_2}{d\gamma}\Big)\frac{d\tau_1}{d\gamma}\right) \tag{22b}
$$

<span id="page-4-4"></span>Introducing Eq. [\(17a\)](#page-3-4) to Eq. [\(22a\)](#page-4-3), a quadratic equation is obtained in terms of  $w_2$ , yielding

$$
\frac{\tau_2 \pi^2}{16R^2 \alpha^2} w_2^2 - \frac{\tau_1}{R} w_2 + \frac{\Gamma_3 \pi^2 \Phi_1}{\Gamma_1 R^2 \alpha^2} - \frac{N_T}{\Gamma_1} = 0
$$
\n(23)

<span id="page-4-6"></span>Solving Eq. [\(23\)](#page-4-4) yields two positive roots

$$
w_2 = \frac{8\tau_1 R \alpha^2}{\tau_2 \pi^2} \left[ 1 \pm \sqrt{1 - \frac{\tau_2 \pi^2}{4\tau_1^2 \alpha^2} \left( \frac{\Gamma_3 \pi^2 \Phi_1}{\Gamma_1 R^2 \alpha^2} - \frac{N_T}{\Gamma_1} \right)} \right]
$$
(24)

<span id="page-4-15"></span>and only the smaller positive one satisfies Eq. [\(19b\)](#page-4-5). By substituting Eq. [\(24\)](#page-4-6) into Eq. [\(12\),](#page-3-5) the crown displacement can be calculated as

$$
w_1 = \frac{8\tau_1 R \alpha^2}{\tau_2 \pi^2} \frac{(-\gamma^2 + 2\gamma)}{(\gamma - 1)^2} \left[ 1 \pm \sqrt{1 - \frac{\tau_2 \pi^2}{4\tau_1^2 \alpha^2} \left( \frac{\Gamma_3 \pi^2 \Phi_1}{\Gamma_1 R^2 \alpha^2} - \frac{N_T}{\Gamma_1} \right)} \right]
$$
(25)

<span id="page-4-7"></span>Introducing Eq. [\(22a\)](#page-4-3) into Eq. [\(21a\)](#page-4-1), the pressure is in the form of

$$
P = \frac{\Gamma_3 \pi^2}{bR^3 \alpha^2} \Phi_1 + \frac{\Gamma_3 \pi^4 (\tau_3 - \tau_2 \Phi_1)}{8bR^4 \alpha^4 \tau_1} w_2
$$
 (26)

<span id="page-4-9"></span>Then introducing Eq.  $(24)$  to Eq.  $(26)$  to eliminate  $w_2$ , the pressure is obtained as

$$
P = \frac{\Gamma_3 \pi^2}{bR^3 \alpha^2} \left[ \left( \frac{\tau_3}{\tau_2} \right) \pm \left( \frac{\tau_3}{\tau_2} - \Phi_1 \right) \sqrt{1 - \frac{\tau_2 \pi^2}{4\tau_1^2 \alpha^2} \left( \frac{\Gamma_3 \pi^2 \Phi_1}{\Gamma_1 R^2 \alpha^2} - \frac{N_T}{\Gamma_1} \right)} \right]
$$
(27)

<span id="page-4-8"></span>To identify the practical equilibrium path, it is necessary to take the first derivative of *P* to *γ*, yielding

$$
\frac{\partial P}{\partial \gamma} = 0 \tag{28}
$$

Newton-Raphson iteration will be used to obtain *γ<sub>cr</sub>* in Eq. [\(28\).](#page-4-8) The specific iteration is as follows: defining an initial  $\gamma_0$ , then substituting  $\gamma_0$ into Eq. [\(28\)](#page-4-8), if  $\left|\frac{\partial P}{\partial \gamma}\right| \ge 0.0001$ , then define  $\gamma_1 = \gamma_0 - 0.001$ , if  $\frac{\partial P}{\partial \gamma}$  ≥ 0.0001, then define  $\gamma_2 = \gamma_1 - 0.001...$  repeating the above process

<span id="page-4-14"></span>until to find  $\left|\frac{\partial P}{\partial y}\right|$  < 0.0001 and its corresponding  $\gamma_{cr}$ . With the substitution of  $\gamma_{cr}$  into Eq. [\(27\)](#page-4-9), the critical buckling pressure for the clampedclamped arches is

$$
P_{cr} = \frac{\Gamma_3 \pi^2}{bR^3 \alpha^2} \left[ \left( \frac{\tau_{3cr}}{\tau_{2cr}} \right) \pm \left( \frac{\tau_{3cr}}{\tau_{2cr}} - \Phi_{1cr} \right) \sqrt{1 - \frac{\tau_{2cr} \pi^2}{4\tau_{1cr}^2 \alpha^2} \left( \frac{\Gamma_3 \pi^2 \Phi_{1cr}}{\Gamma_1 R^2 \alpha^2} - \frac{N_T}{\Gamma_1} \right)} \right]
$$
(29)

where  $\tau_{1cr}$ ,  $\tau_{2cr}$ ,  $\tau_{3cr}$  and  $\Phi_{1cr}$  correspond to  $\gamma = \gamma_{cr}$ .

#### 3.2. Pinned-pinned ends arch

<span id="page-4-10"></span>Taking the first derivative of the energy function Eq. [\(16\)](#page-3-3) for the pinned-pinned case to the two unknowns  $w_2$  and  $\gamma$ , the two equilibrium equations in Eq. (21) are revised into

$$
\left(1 - \frac{w_2 \pi^2}{16R\alpha^2} \frac{2\tau_2}{\tau_1}\right) \bar{N} + \frac{\pi}{2R\alpha^2} \frac{\tau_4}{\tau_1} \left[ (\Gamma_2 \bar{\varepsilon} + M_T) + \Gamma_2 w_2 \frac{\partial \bar{\varepsilon}}{\partial w_2} \right] + \frac{\Gamma_3 w_2 \pi^4}{16R^3 \alpha^4} \frac{2\tau_3}{\tau_1}
$$
\n
$$
= PRb \tag{30a}
$$

<span id="page-4-11"></span>
$$
\left(1 - \frac{w_2 \pi^2}{16R\alpha^2} \frac{d\tau_2}{d\gamma} / \frac{d\tau_1}{d\gamma} \right) \bar{N} + \frac{\pi}{2R\alpha^2} \left[ (\Gamma_2 \bar{\varepsilon} + M_T) \frac{d\tau_4}{d\gamma} + \Gamma_2 \tau_4 \frac{\partial \bar{\varepsilon}}{\partial \gamma} \right]
$$
  

$$
/ \frac{d\tau_1}{d\gamma} + \frac{\Gamma_3 w_2 \pi^4}{16R^3 \alpha^4} \frac{d\tau_3}{d\gamma} / \frac{d\tau_1}{d\gamma} = PRb
$$
 (30b)

<span id="page-4-12"></span>Substituting Eq. [\(30a\)](#page-4-10) into [\(30b\),](#page-4-11) we get a cubic formula in terms of *w*2, yielding

$$
\frac{\pi^2}{16R^2\alpha^2}w_2^3 - \left[\frac{\bar{r}_1}{R} + \frac{\pi\Gamma_2\Phi_2}{2R^2\alpha^2\Gamma_1\tau_2} \left(\frac{\tau_4}{\Phi_2} + 1\right)\right] \nw_2^2 + \left(\frac{\Gamma_3\pi^2\Phi_1}{\Gamma_1R^2\tau_2\alpha^2} + \frac{8\Gamma_2\tau_1\Phi_2}{\Gamma_1R\tau_2\pi} - \frac{N_T}{\Gamma_1\tau_2}\right)w_2 + \frac{8\Phi_2M_T}{\Gamma_1\tau_2\pi} = 0
$$
\n(31)

<span id="page-4-16"></span>where

$$
\Phi_2 = \left(\frac{\tau_4}{\tau_1} - \frac{d\tau_4}{d\gamma} / \frac{d\tau_1}{d\gamma}\right) / \left(\frac{2\tau_2}{\tau_1} - \frac{d\tau_2}{d\gamma} / \frac{d\tau_1}{d\gamma}\right) \tag{32}
$$

For a given value of  $\gamma$ , we get three roots of  $w_{2cr}$  in Eq. [\(31\)](#page-4-12), and only the minimum positive root,  $w_{2cr}$  satisfies Eq. [\(19b\)](#page-4-5). With the Newton-Raphson iteration algorithm, the critical  $\gamma_{cr}$  is obtained in Eq. [\(31\).](#page-4-12) Then substituting  $w_{2cr}$  and  $\gamma_{cr}$  in Eq. [\(30a\),](#page-4-10) the buckling pressure is calculated as

<span id="page-4-13"></span>*P cr*

$$
= \frac{\Gamma_1}{b} \frac{w_{2cr}}{R^2 \tau_{\text{Lcr}}} \left( 1 - \frac{3w_{2cr} \pi^2}{16R\alpha^2} \frac{z_{2cr}}{\tau_{\text{Lcr}}} + \frac{w_{2cr}^2 \pi^4}{128R^2 \alpha^4} \frac{z_{2cr}^2}{\tau_{\text{Lcr}}} \right) + \frac{N_T}{b}
$$
  

$$
\left( \frac{1}{R} - \frac{w_{2cr} \pi^2}{8R^2 \alpha^2} \frac{z_{2cr}}{\tau_{1}cr} \right) + \frac{\Gamma_2}{b} \frac{\pi}{\alpha^2} \frac{\tau_{4cr}}{\tau_{\text{Icr}}} + \frac{N_T}{b \alpha^2} \frac{\tau_{4cr}}{\tau_{\text{Icr}}} + \frac{N_T}{2R^2 \tau_{\text{Icr}}} + \frac{N_T}{2R^2 \tau_{\text{Icr}}} + \frac{N_T}{2R^2 \tau_{\text{Icr}}} + \frac{\Gamma_3}{b} \frac{w_{2cr} \pi^4}{8R^4 \alpha^4} \frac{\tau_{3cr}}{\tau_{\text{Icr}}} \tag{33}
$$

where  $\tau_{1cr}$ ,  $\tau_{2cr}$ ,  $\tau_{3cr}$  and  $\tau_{4cr}$  correspond to  $\gamma = \gamma_{cr}$ . Noting that the pressure capacity  $(P_{cr})$  is independent of the arch width  $(b)$  after substituting Eq. (18) into Eq. [\(33\)](#page-4-13). Accordingly, the unity width  $(b = 1)$  and plane stress condition are employed to improve the computational efficiency in the following numerical simulations.

#### 4. Numerical analysis

#### 4.1. Definition of the numerical model

The instability mechanism of the heated FGM arches under uniform pressure is evaluated by the software ABAQUS [\[35\].](#page-16-16) Following Refs. [\[36,37\],](#page-16-17) the ceramic material of the outside layer is Silicon Nitride  $(Si<sub>3</sub>N<sub>4</sub>)$  and the metal material of the inside layer is Stainless steel (SUS304), respectively. The material properties can be expressed

<span id="page-5-0"></span>Table 1 Material properties of SUS304 and Si<sub>3</sub>N<sub>4</sub>.

| Material                       | Properties          | $\Lambda_{-1}$ | $\Lambda_0$             | 7 Y 1                  | $\Lambda_2$             | $\Lambda_3$              |
|--------------------------------|---------------------|----------------|-------------------------|------------------------|-------------------------|--------------------------|
| <b>SUS304</b>                  | $\lambda_m(K^{-1})$ |                | $12.33 \times 10^{-6}$  | $8.086 \times 10^{-4}$ |                         | O                        |
|                                | $E_m(Pa)$           |                | $210.04 \times 10^{9}$  | $3.079 \times 10^{-4}$ | $-6.534 \times 10^{-7}$ | 0                        |
| Si <sub>3</sub> N <sub>4</sub> | $\lambda_c(K^{-1})$ |                | $5.8723 \times 10^{-6}$ | $9.095 \times 10^{-4}$ |                         |                          |
|                                | $E_c(Pa)$           |                | $348.43 \times 10^{9}$  | $-3.07 \times 10^{-4}$ | $2.16 \times 10^{-7}$   | $-8.946 \times 10^{-11}$ |

<span id="page-5-1"></span>

Fig. 7. The variation of averaged Young's Modulus and coefficient of thermal expansion with temperature.

quantitatively into

$$
\Lambda = \Lambda_0 (\Lambda_{-1} T^{-1} + 1 + \Lambda_1 T + \Lambda_2 T^2 + \Lambda_3 T^3)
$$
\n(34)

where  $\Lambda_{-1}$ ,  $\Lambda_0$ ,  $\Lambda_1$ ,  $\Lambda_2$  and  $\Lambda_3$  are constant as shown in [Table 1](#page-5-0). Here, the averaged Young's modulus  $(\bar{E})$ , as well as the coefficient of thermal expansion  $(\bar{\lambda})$  are integrated as

$$
\bar{E} = \int_{-t/2}^{t/2} bE(z)dz / \int_{-t/2}^{t/2} bdz = [E_m + (E_c - E_m)/(n+1)]
$$
 (35a)

$$
\bar{\lambda} = \int_{-t/2}^{t/2} b\lambda(z) dz / \int_{-t/2}^{t/2} b dz = [\lambda_m + (\lambda_c - \lambda_m)/(n+1)]
$$
 (35b)

and plotted in [Fig. 7](#page-5-1). One may state that  $\bar{\lambda}$  is linear with temperature when  $n$  increases from 0 to 5. On the other hand,  $\bar{E}$  reduces with temperature except for the case  $n = 5$ , where  $\overline{E}$  increases first to a maximum level at  $T = 235.6K$ , and then decreases with the temperature rise. It is easy to implement the definition of Young's modulus and coefficient of thermal expansion by turning on the switch "use temperature dependent data" in Property module. Poisson's ratio is 0.28 for simplicity [\[36\].](#page-16-17)

The geometric parameter is defined as: the radius  $R = 1$ m and the thickness  $t = 0.005$ m, so we get  $R/t = 200$ . For the shallow arch, the central angle  $\alpha$  varies from  $\pi/12$  to  $\pi/4$ . In the thickness direction, the arch is divided evenly into twenty layers  $(N = 20)$ . The material properties are the same in each layer and varies from one layer to another. The effect of layer number on the buckling pressure will be examined later. The arch is discrete with 9600 *CPS*8*R* elements for the case  $R/t = 200$  and  $\alpha = \pi/12$  as shown in [Fig. 8](#page-5-2)(a), where *CPS8R* represents eight-node reduced-integration plane stress elements. For the clamped-clamped arch, the two ends are fixed without any displacements or rotations. For the pinned-pinned arch, the two ends rotate freely as shown in [Fig. 8\(](#page-5-2)b). The ambient temperature field (i.e.

 $T_{amb}$  = 293.15 *K*) and the expected temperature *T* are defined in the first step and the second step of the Abaqus model. Therefore, the temperature rise ( $\Delta T = T - T_{amb}$ ) generates. After that, the uniform pressure is applied at the outside surface as shown in [Fig. 8](#page-5-2)(b).

In addition, the geometric nonlinearities are introduced by switching on the Nlgeom (Non-linear-geometry) option in the Step module of the numerical model. The modified arc-length algorithm is taken to trace the load-displacement path [38–[40\]](#page-16-18) since large displacements or rotations may occur in the present simulation. Only four parameters are defined, including the initial, minimum, maximum increments, and the initial reference loading in the first step. The subsequent pressure value and the displacement size are calculated automatically to track the whole equilibrium paths in the following steps.

#### 4.2. Comparisons and discussions

The numerical results are compared with the analytical solutions of Eq. [\(29\)](#page-4-14) for clamped-clamped arches and Eq. [\(33\)](#page-4-13) for pinned-pinned arches as shown in [Table 2.](#page-6-0) One may observe the numerical buckling pressure are in good agreement with the analytical solutions, and the maximum difference is no more than 5% for all examined cases with two temperature rises (Δ*T*), four different central angles (*α*) and five different volume fraction exponents (*n*), respectively. Moreover, a higher value of volume fraction exponent results in a lower critical buckling pressure except for a few pinned-pinned cases with  $n = 5$  and  $\Delta T = 400K$ . This is because a higher thermal displacement ( $w_T$  in [Fig. 3](#page-1-2)(a)) generates due to the more non-symmetrical distribution of the material properties to the mid-surface of the pinned-pinned arch with  $n = 5$  when  $\Delta T = 400K$ . Such displacement ( $w_T$ ) contributes more to the critical buckling pressure than the effects of the modulus reduction induced by temperature rise.

<span id="page-5-2"></span>

Fig. 8. (a) Mesh, and (b) boundaries of a typical FGM arch.

#### <span id="page-6-0"></span>Table 2

The comparison of critical buckling pressure between present analytical and numerical results.

| $\Delta T$ | α         | n                       | Clamped-clamped (MPa)<br>difference |                       |   | Pinned-pinned (MPa) difference |                       |                             |  |
|------------|-----------|-------------------------|-------------------------------------|-----------------------|---|--------------------------------|-----------------------|-----------------------------|--|
|            |           |                         | <b>FEM</b><br>$(P_e)$               | Eq. $(28)$<br>$(P_a)$ | $\mathcal{P}_a-\mathcal{P}_e$<br>$P_{\alpha}$ | FEM $(P_e)$                    | Eq. $(32)$<br>$(P_a)$ | $P_a - P_e$<br>$P_{\alpha}$ |  |
| 200        | $\pi/12$  | 0                       | 1.97                                | 1.98                  | 0.00  | 1.084                          | 1.100                 | 0.01                        |  |
|            |           | 0.5                     | 1.74                                | 1.74                  | 0.00  | 0.862                          | 0.886                 | 0.03                        |  |
|            |           | 1                       | 1.65                                | 1.66                  | 0.00  | 0.800                          | 0.830                 | 0.04                        |  |
|            |           | $\overline{\mathbf{2}}$ | 1.59                                | 1.59                  | 0.00  | 0.772                          | 0.798                 | 0.03                        |  |
|            |           | 5                       | 1.54                                | 1.52                  | 0.01  | 0.772                          | 0.784                 | 0.02                        |  |
|            | $5\pi/36$ | 0                       | 0.711                               | 0.711                 | 0.00  | 0.383                          | 0.392                 | 0.02                        |  |
|            |           | 0.5                     | 0.623                               | 0.622                 | 0.00  | 0.297                          | 0.308                 | 0.03                        |  |
|            |           | 1                       | 0.591                               | 0.591                 | 0.00  | 0.276                          | 0.286                 | 0.04                        |  |
|            |           | $\overline{2}$          | 0.568                               | 0.567                 | 0.00  | 0.266                          | 0.275                 | 0.03                        |  |
|            |           | 5                       | 0.546                               | 0.540                 | 0.01  | 0.262                          | 0.269                 | 0.02                        |  |
|            | $7\pi/36$ | 0                       | 0.367                               | 0.362                 | 0.02  | 0.193                          | 0.199                 | 0.03                        |  |
|            |           | 0.5                     | 0.321                               | 0.316                 | 0.02  | 0.151                          | 0.155                 | 0.03                        |  |
|            |           | $\mathbf{1}$            | 0.305                               | 0.300                 | 0.01  | 0.139                          | 0.144                 | 0.04                        |  |
|            |           | $\overline{2}$          | 0.292                               | 0.288                 | 0.02  | 0.133                          | 0.138                 | 0.04                        |  |
|            |           | 5                       | 0.281                               | 0.274                 | 0.03  | 0.132                          | 0.135                 | 0.02                        |  |
|            | $\pi/4$   | 0                       | 0.228                               | 0.219                 | 0.04  | 0.116                          | 0.120                 | 0.04                        |  |
|            |           | 0.5                     | 0.199                               | 0.191                 | 0.05  | 0.0913                         | 0.0938                | 0.03                        |  |
|            |           | $\mathbf{1}$            | 0.188                               | 0.181                 | 0.04  | 0.0829                         | 0.0868                | 0.05                        |  |
|            |           | $\overline{2}$          | 0.182                               | 0.174                 | 0.05  | 0.0799                         | 0.0828                | 0.04                        |  |
|            |           | 5                       | 0.174                               | 0.165                 | 0.05  | 0.0791                         | 0.0815                | 0.03                        |  |
| 400        | $\pi/12$  | 0                       | 1.99                                | 2.06                  | 0.03  | 1.115                          | 1.154                 | 0.03                        |  |
|            |           | 0.5                     | 1.74                                | 1.78                  | 0.02  | 0.794                          | 0.839                 | 0.05                        |  |
|            |           | 1                       | 1.65                                | 1.68                  | 0.02  | 0.728                          | 0.764                 | 0.05                        |  |
|            |           | $\overline{2}$          | 1.59                                | 1.60                  | 0.01  | 0.692                          | 0.730                 | 0.05                        |  |
|            |           | 5                       | 1.53                                | 1.52                  | 0.01  | 0.709                          | 0.731                 | 0.03                        |  |
|            | $5\pi/36$ | 0                       | 0.701                               | 0.726                 | 0.03  | 0.383                          | 0.402                 | 0.05                        |  |
|            |           | 0.5                     | 0.607                               | 0.619                 | 0.02  | 0.263                          | 0.278                 | 0.05                        |  |
|            |           | $\mathbf{1}$            | 0.571                               | 0.582                 | 0.02  | 0.237                          | 0.249                 | 0.05                        |  |
|            |           | $\overline{2}$          | 0.546                               | 0.554                 | 0.01  | 0.225                          | 0.236                 | 0.05                        |  |
|            |           | 5                       | 0.523                               | 0.521                 | 0.00  | 0.227                          | 0.236                 | 0.04                        |  |
|            | $7\pi/36$ | 0                       | 0.359                               | 0.367                 | 0.02  | 0.193                          | 0.203                 | 0.05                        |  |
|            |           | 0.5                     | 0.310                               | 0.312                 | 0.01  | 0.129                          | 0.136                 | 0.05                        |  |
|            |           | 1                       | 0.291                               | 0.293                 | 0.01  | 0.117                          | 0.123                 | 0.05                        |  |
|            |           | $\overline{2}$          | 0.278                               | 0.278                 | 0.00  | 0.109                          | 0.115                 | 0.05                        |  |
|            |           | 5                       | 0.266                               | 0.262                 | 0.02  | 0.110                          | 0.116                 | 0.05                        |  |
|            | $\pi/4$   | 0                       | 0.222                               | 0.221                 | 0.00  | 0.116                          | 0.121                 | 0.04                        |  |
|            |           | 0.5                     | 0.192                               | 0.188                 | 0.02  | 0.0780                         | 0.0824                | 0.05                        |  |
|            |           | $\mathbf{1}$            | 0.180                               | 0.176                 | 0.02  | 0.0696                         | 0.0735                | 0.05                        |  |
|            |           | $\overline{2}$          | 0.173                               | 0.167                 | 0.03  | 0.0658                         | 0.0693                | 0.05                        |  |
|            |           | 5                       | 0.164                               | 0.157                 | 0.04  | 0.0668                         | 0.0694                | 0.04                        |  |
|            |           |                         |                                     |                       |   |                                |                       |                             |  |

[Fig. 9](#page-6-1) emphasizes the effects of layer number  $(N)$  on the pressure capacity (buckling pressure). The central angle  $\alpha = \pi/12$  and aspect ratio  $R/t = 200$ . For both boundary conditions examined, the variation of the buckling pressure is negligible when  $N \ge 20$ . Therefore,  $N = 20$  is recommended in industrial manufacturing of the thin-walled FGM

structures.

[Figs. 10 and 11](#page-7-0) depict the comparison of analytical and numerical pressure-displacement equilibrium curves. For clamped-clamped arches, the analytical equilibrium curves are obtained by combining  $w_1 = w_1(\gamma)$  in Eq. [\(25\)](#page-4-15) and  $P = P(\gamma)$  in Eq. [\(27\)](#page-4-9), while for pinnedpinned arches, the curves are obtained by combining  $w_2 = w_2(\gamma)$  in Eq. [\(31\)](#page-4-12) and  $P = P(y)$  in Eq. [\(33\)](#page-4-13), respectively. Noting the crown displacement  $w_1(y)$  can be calculated by substituting the root of Eq. [\(31\)](#page-4-12) into Eq. [\(12\)](#page-3-5) for the pinned-pinned arches. For comparison, the corre-sponding numerical equilibrium paths are also replotted in [Figs. 10 and](#page-7-0) [11,](#page-7-0) respectively. These numerical equilibrium paths show multiple maximum and minimum pressure, which will be explained later in next section. For the two boundary cases, the analytical equilibrium paths agree well with the numerical equilibrium paths both in pre- and postbuckling stages. However, the analytical equilibrium curves only obtain their paths from  $a \rightarrow b \rightarrow c$ , after point *c*, the equilibrium paths are unavailable from the analytical solution. This is because the displacement function of Eq. (11) is based on the symmetrical snap-through buckling deformation, and other deformations are not considered in Eq. (11). In fact, the equilibrium paths  $a \rightarrow b \rightarrow c$  in [Figs. 10 and 11](#page-7-0) are enough since the post-buckling stages after point *b* are unstable and may be avoided in engineering applications.

To verify the validity and consistency of the present analytical and numerical results, a comparison study is performed with the result obtained by Pi and Bradford  $[22]$  for the homogeneous arch  $(n = 0)$ under external pressure and temperature loading. The arch material is ceramic, and the material properties depends on Eq. [\(33\)](#page-4-13) with Young's modulus *Ec* = 348, 430MPa, and coefficient of thermal expansion  $\lambda_c = 5.8723 \times 10^{-6} K^{-1}$  when *T* = 0. The temperature increases from the ambient temperature  $T_{amb}$  to *T*, where  $T_{amb} = 293.15K$ , and  $T = 493.15K$ and 693.15*K*, respectively. The buckling pressure of the present analytical and numerical results, as well as the prediction proposed by Pi and Bradford [\[22\],](#page-16-10) are depicted in [Fig. 12](#page-7-1). For two temperature rises  $\Delta T = 200K$  and 400K, the present analytical and numerical results are in good accordance with the prediction from Pi and Bradford [\[22\]](#page-16-10) for both clamped-clamped and pinned-pinned arches. Therefore, the present numerical results are accurate and reliable to take the parametric study in the next section.

#### 4.3. Parametric evaluations

Some parameters are evaluated and the numerical results are depicted in [Figs. 13](#page-8-0)–28. Unless otherwise mentioned, the central angle  $\alpha = \pi/12$ , and the aspect ratio  $R/t = 200$ . The ambient temperature is  $T_{amb} = 293.15K$ . [Figs. 13 and 14](#page-8-0) depict the effects of volume fraction exponent ( $n = 0, 0.5, 2, 5$ ) on the pressure capacity when  $\Delta T = 200K$ . The pressure capacity decreases with the increase of volume fraction exponent (*n*). This is because Young's modulus reduces when *n*

<span id="page-6-1"></span>

Fig. 9. Effect of layer number (*N*) on the pressure capacity of the FGM arches.

<span id="page-7-0"></span>

Fig. 10. Comparisons of equilibrium paths between analytical and numerical results for clamped-clamped arches.



Fig. 11. Comparisons of equilibrium paths between analytical and numerical results for pinned-pinned arches.

increases as shown in Eq. (35a). Multiple equilibrium paths are observed for both clamped-clamped and pinned-pinned arches, respectively. Similar to the multiple equilibrium paths observed in the homogeneous arch as reported in Ref. [\[41\],](#page-16-19) the FGM arch also shows equilibrium paths. For example, one typical multiple equilibrium path consists of three upper pressure limit (the maximum pressure) points, two lower pressure limit (the minimum pressure) points, and three displacement limit (the maximum and minimum displacements) points respectively, as shown in [Fig. 13\(](#page-8-0)a). The arch reaches its first buckling at point b, beyond which, the arch shows its unstable post-buckling behavior with large deformations. Therefore, the stages after point b are meaningless and may be avoided in engineering applications.

<span id="page-7-1"></span>The equilibrium paths with different temperature rises  $(\Delta T = 100K, 200K, 300K, 400K)$  are plotted in [Figs. 15 and 16](#page-9-0) for both clamped-clamped and pinned-pinned arches, respectively. In all cases,

the displacement starts from a negative value in the horizontal axis due to the thermal radially-outward displacement. Furthermore, the buckling pressure of the clamped-clamped arches in [Fig. 15](#page-9-0) is much higher than that in [Fig. 16](#page-9-1) for pinned-pinned arches, indicating the boundary condition exhibits a significant effect on the buckling pressure. In addition, the buckling pressure varies slightly with the temperature rise because the temperature rise shows two effects on the arch: on one hand, the temperature rise increases the radially-outward displacement, which is beneficial to the buckling pressure; on the other hand, the temperature rise reduces Young's modulus, which is unbeneficial to the buckling pressure. Therefore, the buckling pressure may be nonlinear with the temperature rise.

[Figs. 17 and 18](#page-10-0) show the distributions of bending moment through the arch span for clamped-clamped and pinned-pinned arches, respectively. All the values correspond to the critical buckling stage. For both



Fig. 12. Comparisons of buckling pressure between the present analytical and numerical results, as well as prediction from Pi and Bradford [\[22\]](#page-16-10) for (a)  $\Delta T = 200K$ and (b)  $\Delta T = 400K$ .

<span id="page-8-0"></span>

Fig. 13. The equilibrium paths of clamped-clamped FGM arches with different volume fraction exponents (n).

boundary conditions, the maximum positive bending moment occurs on the crown (mid-span), while the maximum negative bending moment occurs on the position between mid-span and edge, corresponding to the position with the maximum radial displacement. Furthermore, a

higher temperature rise results in an increase in the bending moment. In addition, the bending moment is nonlinear with respect to volume fraction exponent  $(n)$ . This is because the thermal moment generates due to the non-symmetrical distribution of material properties in the



Fig. 14. The equilibrium paths of pinned-pinned FGM arches with different volume fraction exponents (n).

<span id="page-9-0"></span>

Fig. 15. The equilibrium paths of clamped-clamped FGM arches under varied temperature rises.

cross-section of the arch. From Eq. [\(18e\)](#page-3-6), the thermal moment does not vary uniformly with volume fraction exponent. However, the difference of the bending moment is very small when volume fraction exponent varies from 0.5 to 5, for both clamped-clamped and pinned-pinned arches, respectively. The distributions of the hoop force through the arch span are shown in [Figs. 19 and 20](#page-10-1) for clamped-clamped and pinned-pinned arches, respectively. All the values correspond to the critical buckling level. The

<span id="page-9-1"></span>

Fig. 16. The equilibrium paths of pinned-pinned FGM arches under varied temperature rises.

<span id="page-10-0"></span>

Fig. 17. The bending moment of the clamped-clamped arch with different temperature rises.



Fig. 18. The bending moment of the pinned-pinned arch with different temperature rises.

<span id="page-10-1"></span>

Fig. 19. The hoop force of the clamped-clamped arch with different temperature rises.

hoop force decreases with the increase of volume fraction exponent  $(n)$ , and a higher temperature rise results in a lower hoop force. This is because the increase of volume fraction exponent and temperature rise will reduce Young's modulus. A few fluctuations of the hoop force are observed, and these fluctuations are neglected in Eq. (17). The hoop force is simplified as constant. Such a simplification results in an insignificant difference on the buckling pressure as shown in [Table 2.](#page-6-0)

[Figs. 21](#page-11-0)–23 depict the distribution of the hoop strain and stress at the top, bottom and middle surfaces through the span of the clampedclamped arch. At the top surface, the strain and stress are positive near the crown area ( $\phi/\alpha = 0$ ) whereas the hoop strain and stress are negative near the area with the maximum radial displacement as shown in

[Fig. 21\(](#page-11-0)a) and (b). These features indicate that tension occurs at the top surface of the crown areas, while compression occurs at the top surface of the area with the maximum radial displacement. The top surface tensile strain on the crown area increases with increasing volume fraction exponent and temperature. However, the compressive stress and strain near the maximum displacement change slightly when the temperature rise increases from 200*K* to 400*K*. On the other hand, the strain and stress at the bottom surface are negative near the crown area  $(\phi/\alpha = 0)$  whereas the hoop strain and stress are positive near the position with the maximum radial displacement as shown in [Fig. 22.](#page-12-0) The bottom surface tensile strain on the maximum radial displacement area increases with increasing volume fraction exponent and temperature.



Fig. 20. The hoop force of the pinned-pinned arch with different temperature rises.

<span id="page-11-0"></span>

Fig. 21. The distribution of strain and stress at the top surface of the clamped-clamped arch.

Nevertheless, the volume fraction exponent, as well as the temperature rise, show small effects on the crown compressive stress and strain. In the middle surface, it is found the hoop strain is positive whereas the stress is negative and compressive as shown in [Fig. 23](#page-12-1). This is induced by the nature of thermal stresses. It should be noted that the case of  $n = 0$  shows different strain and stress distributions from other cases because the material property of  $n = 0$  is uniform in the cross-section. However, for other cases  $(n = 0.5, 1, 2, 5)$ , additional hoop stress and strain are generated due to the non-symmetrical distribution of the material properties with respect to the middle surface of the FGM arches.

[Figs. 24](#page-13-0)–26 depict the distribution of the hoop strain and stress at the top, bottom and middle surfaces of the pinned-pinned arch. Like the clamped-clamped cases, the maximum tensile strain and stress occur at the crown whereas the maximum compressive strain and stress occur at the maximum radial displacement position of the top surface in [Fig. 24](#page-13-0). On the contrary, the maximum compressive strain and stress occur at the crown whereas the maximum tensile strain and stress occur at the maximum radial displacement position of the bottom surface in [Fig. 25](#page-13-1). At the middle surface, all hoop strains are positive and change slightly through the arch span as shown in [Fig. 26](#page-14-0). The hoop stresses are compressive, which is similar to the clamped-clamped arches.

[Figs. 27 and 28](#page-14-1) illustrate the distribution of hoop strain and stress in the thickness direction of the mid-span ( $\phi/\alpha = 0$ ). All strain and stress correspond to the critical buckling level. It is seen that the hoop strain is distributed linearly in the thickness direction for both clamped-clamped

<span id="page-12-0"></span>

Fig. 22. The distribution of strain and stress at the bottom surface of the clamped-clamped arch.

<span id="page-12-1"></span>

Fig. 23. The distribution of strain and stress at the middle surface of the clamped-clamped arch.

<span id="page-13-0"></span>

Fig. 24. The distribution of strain and stress at the top surface of the pinned-pinned arch.

<span id="page-13-1"></span>

Fig. 25. The distribution of strain and stress at the bottom surface of the pinned-pinned arch.

<span id="page-14-0"></span>

Fig. 26. The distribution of strain and stress at the middle surface of the pinned-pinned arch.

<span id="page-14-1"></span>

Fig. 27. Distribution of mid-span strain and stress in the thickness direction for the clamped-clamped arch.



Fig. 28. Distribution of mid-span strain and stress in the thickness direction for the pinned-pinned arch.

and pinned-pinned arches, respectively. Generally, the higher the temperature, the higher the hoop strain and stress. Similarly, the higher the volume fraction exponent, the higher the hoop strain except for the vicinity of the bottom surface. However, the stress is distributed nonlinearly due to the non-symmetrical dispersion of Young's modulus in the thickness direction  $(n > 0)$ . For a homogeneous arch  $(n = 0)$ , the stress is linear in the thickness direction.

#### 5. Conclusions

The stability mechanism of the FGM arch is evaluated when it is subjected to external pressure and thermal loading. Based on the present analytical and numerical investigation, several main conclusions are emphasized:

- (1) The derived analytical buckling pressure is verified successfully by the numerical results for both clamped-clamped and pinned-pinned arches, respectively. For a homogenous arch  $(n = 0)$ , the present analytical and numerical results are consistent with available other closed-form expressions.
- (2) The strain and stress of the FGM arch  $(n > 0)$  are distributed differently from the homogeneous arch due to the non-symmetrical distribution of the material properties in the cross-section.
- (3) The temperature rise generates a thermal radially-outward displacement and a reduction on Young's modulus, which are positive and negative to the critical buckling pressure of the FGM arch, respectively.
- (4) The layer number  $N = 20$  in the thickness direction is accurate enough to converge the buckling pressure. Therefore,  $N = 20$  is recommended to industrial manufacturing of the thin-walled FGM structures.

The present investigation is mainly concerned with the elastic buckling of the FGM arch. Further experimental verification is necessary before it is applied to engineering applications. In addition, the inelastic material properties are ignored in the present model. However, they may exist in practical systems and have significant effects on the buckling pressure.

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#### Declaration of Competing Interest

The authors declare the present investigation has no conflict of interests with any other works or studies.

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